# TURBULENT MHD FLOW OF A RADIATING GAS

**THEODORE F.** SMITH, FARIBORZ ALIPOUR-HAGHIGHI and PHILLIP H. PAUL\*

Division of Energy Engineering, The University of Iowa,

Iowa City, IA 52242, U.S.A.

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**Abstract-** The high temperatures and gaseous products of combustion of hydrocarbon fuels in proposed large scale magnetohydrodynamic generators have resulted in a need to develop accurate models for prediction of gas temperature profiles and wall heat fluxes. At the high temperatures and large sizes, radiative heat transfer in the combustion gases may be a significant energy transport mechanism. Results are presented for gas temperature profiles for fully developed turbulent magnetohydrodynamic flow of a radiating gray gas between infinite parallel plates with a magnetic field applied perpendicular to the plates. Emphasis is placed on examination of temperature profiles and Nusselt numbers for various values of Hartmann and Reynolds numbers, gas optical thickness, and conduction to radiation parameter.

### NOMENCLATURE

$A_i$	coefficients defined by equation (28);
Ā,	van Driest constant, 26.0;
Б,	Mei and Squire constant, 3.4;
$B_b$ ,	black body emissive power $[W/m^2]$ ;
$B_m$	magnetic field $[Webers/m^2];$
B.	magnetic interaction, $Ha^2/Re$ ;
$c_p$	constant pressure specific heat
	$[kJ/kg-K];$
$C_f,$	friction coefficient;
$C_i$	terms defined by equation (29);
D.	magnetic damping function;
$D_e$	hydraulic diameter [m];
$D_i$	terms defined by equation $(30)$ ;
Е,	electric field $[V/m]$ ;
Ec.	Eckert number, $\bar{v}^2/c_p T_w$ ;
$E_n$	exponential integral of order $n$ ;
$F_1$ ,	function defined by equation (7);
Ha,	Hartmann number, $\delta B_m (\sigma_e/\mu)^{1/2}$ ;
j,	current density $[A/m^2]$ ;
J,	dimensionless current, $j/\sigma_e \bar{v} B_m$ ;
k,	thermal conductivity $[W/m-K];$
K.	power factor, $E/\bar{v}B_m$ ;
$K_{1}, K_{2},$	kernels defined by equations (16) and
	$(20)$ ;
Ŕ,	Von Karman constant, 0.4;
L,	channel width $[m]$ ;
M,	number of nodal points;
N,	conduction to radiation parameter,
	$k\kappa/4\sigma T_w^3$ ;
Nu,	Nusselt number, $D_e q_w / k(T_w - T_b)$ ;
$Pr, Pr_t$	molecular and turbulent Prandtl num-
	bers, $v/\alpha$ , $v_t/\alpha$ ;
q,	heat flux $[W/m^2];$
Q,	dimensionless heat flux, $-q/4\tau_0\sigma T_w^4$ ;
Re,	channel Reynolds number, $v_c \delta/v$ ;
Re <sub>p</sub>	hydraulic Reynolds number, $v_c D_e/v$ ;
Re <sub>t</sub> ,	turbulent Reynolds number, $Re(C_f/2)^{1/2}$ ;

<sup>\*</sup> Presently, Research Assistant, Stanford University.

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- $t$ , parameter;<br> $T$ , temperature
- $T$ , temperature [K];<br>velocity  $\lceil m/s \rceil$ ;
- v, velocity  $[m/s]$ ;<br>v<sub>c</sub>,  $\bar{v}$ , centerline and r  $v_c$ ,  $\bar{v}$ , centerline and mean velocities [m/s];<br>
V. normalized velocity,  $v/\bar{v}$ ;
- $V$ , normalized velocity,  $v/\bar{v}$ ;<br> $V_r$ , scaling factor, 4  $Re_t$ ;
- $V_t$ , scaling factor, 4  $Re_t$ ;<br>x, y, z, coordinates  $[m]$ ;
- $x, y, z$ , coordinates [m];<br> $\bar{z}$ , turbulent coordin
- turbulent coordinate.

# Greek symbols



# **INTRODUCTION**

THE **HIGH** temperatures of gaseous products resulting from combustion of hydrocarbon fuels have resulted in a need to develop more accurate models for prediction of gas temperature profiles and wall heat fluxes in the magnetohydrodynamic generator (MHD). Information concerned with gas temperature profiles particularly near the generator walls is necessary in order to evaluate the gas electrical conductivity and associated wall heat fluxes. Successful design of a wall cooling system and selection of materials for the generator walls also require estimates for wall heat fluxes. At the high operating temperatures and in the presence of combustion products of carbon dioxide and water vapor, particles, as well as a seed material which acts as a plasma, radiative heat transfer is expected to become an important factor as the size of the generator increases  $\lceil 1 \rceil$ . The overall objective of this research is an examination of radiative transfer in MHD flows. Resuits from this study are also expected to find application to gas-cooled nuclear reactors and flow of plasma.

Viskanta  $[2,3]$ , Cramer and Pai  $[4]$  as well as Wilson and Haji-Sheikh [5] examined radiative transfer for MHD flow between infinite parallel plates for gray gas properties where the absorption coefficient is independent of frequency. Gupta and Gupta [6] as well as Datta and Janta  $[7]$  investigated radiative transfer effects for flow of an optically thin, electrically conducting fluid in a vertical channel. Helliwell  $[8,9]$ as well as Helliwell and Mosa  $\lceil 10 \rceil$  utilized the differential approximation  $\lceil 11 \rceil$  to describe radiative transfer in MHD channel flow. Smith and Paul [12] employed a band absorption model to examine radiative transfer in MHD flow. In these investigations, the how was assumed laminar with several of the analyses  $[2, 3, 5, 8-10, 12]$  utilizing velocity profiles as obtained from the Hartmann MHD flow results [ 131. In most MHD generators, however, the flow is expected to be turbulent and results derived from these analyses may not be directly applicable to proposed generators. This concern is illustrated by the fact that a Hartmann number of 200 which is representative of that for the U-25 channel [14] yields physically unreasonably high channel centerline temperatures when inserted into the above analyses. Consequently. a need exists to provide results concerned with radiative transfer in turbulent MHD flows.

The purpose of this paper is to present results to illustrate effects of radiative transfer in turbulent MHD flow. The system selected for study as depicted in Fig. 1 consists of hydrodynamically and thermally established turbulent steady flow of an electrically conducting and radiatively participating gas between two infinite parallel plates with an imposed uniform and constant magnetic field applied in the positive  $z$ -



FIG. 1. Schematic diagram of MHD channel.

direction. A constant electric field is applied in the positive y-direction. This system was chosen for convenience of including MHD effects and enabling the radiative transfer contribution to be evaluated. Analyses and results presented here, however. are expected to be applicable to more comprehensive MHD flow models. Channel walls are isothermal and black for radiative transfer purposes. Physical properties of the gas are constant. The gas is in local thermodynamic equilibrium and emits and absorbs thermal radiation with a constant absorption coefficient. The gas has a refractive index of unity, and scattering effects are negligible. Results are sought for gas temperature profiles and Nusselt numbers as functions of system parameters.

#### **ANALYSIS**

#### Energy balance

The thermal energy equation for fully developed turbulent flow of an electrically conducting and radiatively participating gas between infinite parallel plates where viscous dissipation and Joule heating effects are considered but axial components of conductive, convective and radiative transfer are neglected is [13, 15, 16]

$$
\frac{d}{dz} \left[ k \left( 1 + \frac{\alpha_t}{\alpha} \right) \frac{dT}{dz} \right]
$$
\n
$$
= -\mu \left( 1 + \frac{v_t}{v} \right) \left( \frac{dy}{dz} \right)^2 - \frac{j^2}{\sigma_r} + \frac{dq_r}{dz} \quad (1)
$$

where constant properties have been assumed throughout. In equation (l), the LHS represents the net thermal energy transport due to molecular conduction and turbulent transport with  $\alpha_t$  denoting the turbulent diffusivity of heat. The terms on the RHS arc in the order shown the molecular and turbulent viscous dissipation with  $v_t$  being the turbulent viscosity of momentum, Joulean heating with  $\sigma_e$  denoting the gas electrical conductivity, and divergence of the radiative flux. Both the turbulent transport quantities are assumed to vary across the channel. The boundary conditions for the temperature distribution are  $T(0) =$  $T(L) = T_w$ . The following dimensionless quantities are introduced

$$
\eta = \frac{z}{\delta}, \quad \theta = \frac{T}{T_w}, \quad V = \frac{v}{c}.
$$
\n
$$
Pr = \frac{v}{\alpha}, \quad Pr_t = \frac{v_t}{\alpha_t}, \quad Ec = \frac{\tilde{v}^2}{c_p T_w}
$$
\n
$$
Ha = \delta B_m \left(\frac{\sigma_e}{\mu}\right)^{1/2}, \quad J = \frac{j}{\sigma_e \tilde{v} B_m},
$$
\n
$$
N = \frac{k\kappa}{4\sigma T_w^3}, \quad \tau_0 = \kappa L, \quad Q = \frac{-q}{4\tau_0 \sigma T_w^4} \tag{2}
$$

where *Pr* and *Pr,* denote the molecular and turbulent  $\cdot$  Prandtl numbers, respectively. *Ha* is the Hartmann number based on the half-channel width  $\delta$ , and N represents the ratio of molecular thermal conduction

to radiation for a gas with an absorption coefficient  $\kappa$ and optical thickness  $\tau_0$  based on the channel width. Equation (1) thus becomes

$$
\frac{d}{d\eta} \left[ \left( 1 + \frac{Pr}{Pr_t} \frac{v_t}{v} \right) \frac{d\theta}{d\eta} \right] = -PrEc
$$
\n
$$
\times \left[ \left( 1 + \frac{v_t}{v} \right) \left( \frac{dV}{d\eta} \right)^2 + Ha^2 J^2 \right] - \frac{\tau_0^2}{2N} \frac{dQ_t}{d\eta} \quad (3)
$$

with boundary conditions  $\theta(0) = \theta(2) = 1$ . As a result of the numerical scheme employed to solve equation (3), the derivative on the LHS was retained. Since  $Q_r$  is expressed by an integral term with temperature appearing to the fourth power in the integrand, equation (3) constitutes a second-order, non-linear, integrodifferential equation for the gas temperature distribution. In order to predict the gas temperature profiles, expressions for gas velocity profiles, turbulent viscosity, current density, and radiative flux must be specified and are provided in the following sections.

#### *Velocity, turbulent viscosity, and current models*

The flow in several proposed MHD power generators such as the U-25 is subsonic and turbulent. Thus, in considering subsonic flow, the relation of the turbulent boundary layer and free stream is assumed to be similar to that encountered in ordinary hydrodynamic (OHD) flows where magnetic effects are absent. Kruger and Sonju [17] by employing the Karman-Pohlhausen technique have provided an estimate of wall shear stress and boundary-layer thickness corresponding to the semi-empirical velocity correlations proposed by Harris [18]. The local velocity normalized with the centerline value is evaluated from

$$
\frac{v(\eta)}{v_c} = \left(\frac{C_f}{2}\right)^{1/2} \left[ 6.154 + 2.457
$$
  
× ln(*Re<sub>t</sub>* $\eta$ ) + *F*<sub>1</sub> $\left(\frac{Ha^2}{Re_t}\eta\right)$ ]. (4)

Kruger and Sonju [17] noted that for MHD flow over a flat plate, the boundary-layer thickness and wall shear stress approach asymptotic values. Since the present study addresses fully developed flow conditions, the asymptotic value for wall shear stress was employed for the friction coefficient in equation (4). Graphical results for the asymptotic friction coefficient presented by Kruger and Sonju [17] are approximated by

$$
\frac{C_f}{2} = [10.536 + 0.929 \ln(\bar{B})
$$
  
+ 0.0222 ln<sup>2</sup>( $\bar{B}$ )] × 10<sup>-3</sup> (5)

where  $\bar{B}$  (=  $Ha^2/Re^2$ ) is the interaction parameter. The turbulent Reynolds number is defined as

$$
Re_t = Re \left(\frac{C_f}{2}\right)^{1/2} = \frac{Re_p}{4} \left(\frac{C_f}{2}\right)^{1/2} \tag{6}
$$

where  $Re\ (=Re<sub>D</sub>/4)$  and  $Re<sub>D</sub> (=4v<sub>c</sub>\delta/v)$  denote the channel and hydraulic Reynolds numbers, respectively. The function  $F_1(\zeta)$  is presented graphically by Harris [18] and is approximate by the following expression [17] for  $\zeta \le 0.6$ 

$$
F_1(\zeta) = 2.502 + 21.930\zeta
$$
  
-(6.259 + 53.747\zeta + 649.535\zeta^2)<sup>1/2</sup> (7a)

and for  $\zeta > 0.6$ 

$$
F_1(\zeta) = -2.07 - 2.457 \ln(\zeta). \tag{7b}
$$

Near the wall when  $\eta$  is small, velocities evaluated from equation (4) become negative as a result of the logarithmic term. Therefore, in a manner similar to that employed in OHD turbulent flows for the laminar sublayer, velocities are calculated utilizing the product of *Re,v* up to the position where this product equals equation (4). The mean velocity is determined by integration of the local velocities across the halfchannel width. The velocity ratio  $V$  is then formed by dividing the local velocities by the mean velocity. Finally, differentiation of the velocity ratio yields the velocity gradient for insertion into the viscous dissipation term of equation (3).

Expressions for the turbulent viscosity for MHD flow are generally based on those for turbulent viscosity for OHD flow with modifications to account for such factors as the damping of the turbulent viscosity as the magnetic field is increased. For the present study, the OHD turbulent viscosity model of van Driest [19] as modified by the Mei and Squire channel factor [20] is utilized with a multiplicative magnetic damping function utilized by Fiveland [21]. Thus,

$$
\frac{v_t}{v} = \frac{0.5 D}{1 + \overline{b} \eta} \{ [1 + 4 \overline{K}^2 \overline{z}^2 (1 - e^{-\overline{z}/\overline{A}})^2 ]^{1/2} - 1 \}
$$
(8)

where  $\bar{b}$ ,  $\bar{K}$ , and  $\bar{A}$  are the Mei and Squire, Von Karman, and van Driest constants, respectively, and the turbulent distance is

$$
\bar{z} = \eta R e_r. \tag{9}
$$

The magnetic damping function is [21]

$$
D = e^{-700\,Ha^2/Re_t^2}.
$$
 (10)

The OHD turbulent viscosity model has also been employed by Edwards and Balakrishnan [22] in studies of turbulent channel flow of a radiating gas. The turbulent Prandtl number is assumed unity for the results presented here.

The electrically conducting gas moving through the imposed magnetic field produces an induced current flowing in the negative y-direction of Fig. 1. In dimensionless form, the induced current is  $[13, 17]$ 

$$
J(\eta) = K - V(\eta) \tag{11}
$$

where the power factor  $K = E/\bar{v}B_m$  with *E* representing the applied electric field. If  $K = 1$ , an open circuit exists and the net current given by the integration of equation (11) across the channel is zero. A short circuit results in  $E = K = 0$ . Values of  $K < 1$  and  $> 1$  imply, integral, and the dimensionless variables, equation (17) respectively, to an MHD generator and pump. The becomes additional Joulean dissipation due to turbulent fluctuations in the velocity  $[18]$  is neglected  $[21]$ .

#### Radiative transfer model

The formulation of the gas radiative transfer model considers the gas to have a constant absorption coefficient. From Sparrow and Cess [16], the radiative flux for an emitting and absorbing gray gas in local thermodynamic equilibrium is

$$
q_r(\tau) = 2E_b(0)[E_3(\tau) - E_3(\tau_0 - \tau)]
$$
  
+ 2\int\_0^{\tau\_0} E\_b(t)E\_2(|\tau - t|)sgn(\tau - t)dt (12)

where optical thickness is defined as  $\tau = \kappa z$ .  $E_2$  and  $E_3$ denote second and third exponential integrals, respectively.  $E<sub>b</sub>$  is the black body emissive power given by the Stefan -Boltzmann law. The wall radiative flux is needed in the Nusselt number expressions discussed later and is found by evaluating equation (12) at the wall where  $\tau = 0$ . Thus, with  $E_3(0) = 1/2$ 

$$
q_r(0) = 2E_b(0) \left[ \frac{1}{2} - E_3(\tau_0) \right] - 2 \int_0^{\tau_0} E_b(t) E_2(t) dt.
$$
 (13)

Introducing the dimensionless parameters defined by equation (2) and the integral property of the exponential integrals, equation (13) is transformed to

$$
Q_r(0) = \frac{1}{4} \int_0^2 (\theta^4 - 1) E_2\left(\frac{\tau_0}{2}\eta\right) d\eta. \tag{14}
$$

Recognizing that the temperature profiles are symmetric about the channel centerline, wall heat flux may be evaluated from

$$
Q_r(0) = \frac{1}{4} \int_0^1 (\theta^4 - 1) K_1(\eta) d\eta
$$
 (15)

where the kernel  $K_1$  is defined as

$$
K_1(\eta) = E_2\left(\frac{\tau_0}{2}\eta\right) + E_2\left[\frac{\tau_0}{2}(2-\eta)\right].
$$
 (16)

The above form for wall radiative flux is convenient since the numerical integration is carried out only for one-half of the channel.

The divergence of radiative flux needed for solution of equation (3) is found by differentiation of equation (12) with respect to  $\tau$ . Thus,

$$
\frac{dq_r}{d\tau} = 4E_b(\tau) - 2E_b(0)[E_2(\tau) + E_2(\tau_0 - \tau)]
$$

$$
- 2\int_0^{\tau_{\text{th}}} E_b(t)E_1(|\tau - t|)dt \quad (17)
$$

where  $E_1$  is the first exponential integral. Introducing the technique presented by Mingle [23] for removal of the singularity in the integral of the first exponential

$$
\frac{dQ_r}{d\eta} = \frac{\tau_0}{8} \int_0^2 \left[ \theta^4(\zeta) - \theta^4(\eta) \right]
$$

$$
\times E_1 \left[ \frac{\tau_0}{2} (|\eta - \zeta|) \right] d\zeta - \frac{1}{4} \left[ \theta^4(\eta) - 1 \right] K_1(\eta). \quad (18)
$$

Finally, employing the symmetric temperature profile conditions, the divergence of the radiative flux is

$$
\frac{dQ_r}{d\eta} = \frac{\tau_0}{8} \int_0^1 \left[ \theta^4(\zeta) - \theta^4(\eta) \right]
$$

$$
\times K_2(\eta, \zeta) d\zeta - \frac{1}{4} \left[ \theta^4(\eta) - 1 \right] K_1(\eta) \quad (19)
$$

where the kernel  $K_2$  is defined as

$$
K_2(\eta,\zeta) = E_1\left[\frac{\tau_0}{2}(|\eta-\zeta|)\right] + E_1\left[\frac{\tau_0}{2}(2-\zeta-\eta)\right].
$$
 (20)

Hence, only information about temperature profiles in the half-channel is needed.

# *Wail heat jlux and Nusselt numbers*

The wall heat flux is the sum of the conductive and radiative fluxes evaluated at the wall and is given by

$$
q_w = -k \frac{dT}{dz} \bigg|_{z=0} + q_r(0) \tag{21}
$$

which may be expressed as

$$
Q_w = \frac{N}{\tau_0^2} \left. \frac{d\theta}{d\eta} \right|_{\eta=0} + Q_r(0). \tag{22}
$$

Another quantity of interest is the Nusselt number defined in terms of the wafl heat flux and hydraulic diameter as

$$
Nu = \frac{D_e q_w}{k(T_w - T_b)}\tag{23}
$$

where  $T<sub>b</sub>$  is the bulk fluid temperature. Equation (22) in conjunction with equation (23) transform to

$$
Nu = Nu_c + Nu_r \tag{24}
$$

where convective and radiative Nusselt numbers are defined respectively as

$$
Nu_c = \frac{4}{\theta_b - 1} \left. \frac{d\theta}{d\eta} \right|_{\eta = 0};
$$
  

$$
Nu_r = \frac{4\tau_0^2}{(\theta_b - 1)N} Q_r(0). \quad (25)
$$

The bulk fluid temperature is found from

$$
\theta_b = \int_0^1 \theta V \, \mathrm{d}\eta. \tag{26}
$$

The above definition of the convective Nusselt number includes effects attributed to radiative transfer since the gas temperature gradient at the wall is partially appearing in the analysis were evaluated utilizing a governed by the radiant exchange process. variable spacing trapezoidal integration algorithm.

# **METHOD OF SOLUTION**

Solutions of the energy equation were arrived at by a finite difference scheme where equation (3) was replaced with a set of linear algebraic equations with the source and nonlinear terms on the RHS of equation (3) assumed known. An iterative technique was then utilized since the radiative term varies with temperature. The finite difference scheme for symmetry conditions about the channel centerline yields the following set of algebraic equations for the temperatures

$$
A_{i-1}\theta_{i-1} - (A_{i-1} + A_i)\theta_i + A_i\theta_{i+1} = -C_i - D_1;
$$
  
\n
$$
i = 2 \text{ to } M - 1 \quad (27a)
$$
  
\n
$$
2A_{M-1}(\theta_{M-1} - \theta_M) = -C_M - D_M \quad (27b)
$$

where  $i = 1$  and M correspond to positions at the channel wall and centerline, respectively, and  $\theta_1 = 1.0$ . The coefficients of  $\theta_i$  are evaluated from

$$
A_{i} = \left(1 + \frac{Pr}{Pr_{i}} \frac{v_{t}}{v}\right)_{i+1/2} / (\eta_{i+1} - \eta_{i});
$$
  
  $i = 1$  to  $M - 1$  (28)

where the subscript  $i + 1/2$  denotes that the quantity is to be evaluated at the midpoint between adjacent nodes. Also,  $\eta_1 = 0$ , and  $\eta_M = 1.0$ . The temperature independent source terms of viscous dissipation and Joulean heating are represented by  $C_i$  which are determined from

$$
C_i = \Delta \eta_i PrEc \left[ \left( 1 + \frac{v_t}{v} \right) \left( \frac{dV}{d\eta} \right)^2 + Ha^2 J^2 \right]_i;
$$
  
  $i = 2$  to M (29)

where  $\Delta \eta_i$  is the width of each node. Nodal boundaries are placed at the midpoint between adjacent nodes. The temperature dependent radiative terms described by  $D_i$  are given as

$$
D_i = \Delta \eta_i \frac{\tau_0^2}{2N} \left. \frac{\mathrm{d}Q_r}{\mathrm{d}\eta} \right|_i; \quad i = 2 \text{ to } M \tag{30}
$$

With the source terms known, equation (27) was solved utilizing a tridiagonal matrix algorithm. The iteration scheme consisted of specification of an initial temperature distribution to evaluate the radiative term. New temperatures from equation (27) were then obtained. If the new temperatures upon comparison with the previous temperatures do not satisfy a convergence error criterion, the previous temperatures were adjusted with a relaxation factor multiplied by the difference between the new and previous temperatures and the iteration process repeated. The relaxation factor whose value was typically 0.1 was employed to ensure that nonlinearities attributed to the radiative term were suppressed and that convergence could be obtained. Upon achievement of convergence, the bulk fluid temperature, wall heat fluxes, and Nusselt numbers were calculated. Integrals

For turbulent flows, velocity and temperature profiles vary significantly near the channel wall and a nonuniform grid spacing must be employed to maintain a manageable number of grid points. The nodal positions were assigned according to the following expression [22]

$$
\eta = \frac{1}{V_t} (e^{\xi} - 1); \quad \xi = \ln(\eta V_t + 1) \tag{31}
$$

where  $V_t$  (=4 $Re_t$ ) is a scaling factor whose purpose is to place more points in the vicinity of the wall. The constant utilized in  $V_t$  was selected after some numerical experiments were performed to establish the influence of nodal positions on the temperature distributions. The distance  $\xi_c$  as measured from the wall to the channel center where  $\eta = 1$  in equation (31) is subdivided into  $M-1$  equally spaced nodes. Equation (31) then yields a variable grid spacing for values of  $\eta$ with sufficient number of points near the channel wall. For the results presented here,  $M = 51$  was found to provide accurate results within reasonable computational times.

#### **RESULTS AND DISCUSSION**

#### *Parameter values*

Before results from the analysis are presented, it is informative to identify physically representative values of the dimensionless quantities which govern the solutions. Values for the dimensional parameters were selected as typical of those for a large scale MHD generator such as the U-25 channel [14]. For this channel, the mean velocity is approximately 900 m/s, centerline temperatures are near 28OOK, wall temperatures may range from 1000 to 1800 K depending on the wall cooling system, electrode materials, and presence of a molten slag layer, channel widths vary from 0.35 to 0.75 m, the magnetic field is near 2 Webers/m', and the gas has an average electrical conductivity of  $10 \Omega^{-1}$  m<sup>-1</sup>. Utilizing these values as well as gas property data taken from Bunde [24], representative values for the dimensionless parameters are  $Re_p = 3 \times 10^6$ ,  $Ha = 200$ ,  $Ec = 0.2$ , and  $N =$ 0.0001 to 0.01 with  $\tau_0 = 0.1$  to 10. Definitive values for  $\tau_0$  are unavailable; however, Wilson and Haji-Sheikh [5] quoted an absorption coefficient of 0.0001 m<sup>-1</sup> for a small scale MHD channel, and Smith and Paul [12] reported optical thicknesses ranging from 1 to 10 for a gas composition similar to that in the U-25 channel. In selection of values for the governing parameters, cognizance was made of these approximate but physically meaningful values of the parameters,

# *Transparent gas*

Representative velocity and temperature profiles for a transparent gas displayed in Figs. 2 and 3, respectively, are intended to illustrate the turbulent characteristics and behavior of the results. Profiles are shown for OHD and MHD flows with hydraulic



Flc,. 2. Velocity profiles



FIG. 3. Temperature profiles for a transparent gas.



**FIG.** 4. Temperature profiles as a function of optical thickness.

Reynolds numbers of  $1 \times 10^6$  and  $3 \times 10^6$  and Hartmann numbers of zero for the OHD results and 150 and *200* for the MHD flows. The OHD velocity profiles are based on the universal turbulent profiles with the coefficient of friction evaluated from the Filonenko expression [25]. The OHD velocity profiles exhibit smaller gradients near the wall and lower mean velocities than the MHD velocity profile for the same centerline velocities. As the Hartmann increases the profiles become more uniform with higher gradients near the wall. For the OHD velocity profiles, higher Reynolds numbers produce flatter profiles and steeper velocity gradients near the wall. However, for the MHD velocity profiles, higher Reynolds numbers produce the opposite effect as displayed by results for *Ha =* 200. As presently modeled, the magnetic damping function in the turbulent viscosity expression does not influence the velocity profiles. The OHD temperature profiles in Fig. 3 are the result of viscous dissipation and exhibit values near 1.07 for the two Reynolds numbers. The inclusion of the Joulean heating term in the energy equation is seen to produce significantly higher gas temperatures for MHD flows than for the OHD case. Values of the Hartmann number of 150 and 200 produce, respectively, centerline temperatures of 2.4 and 3.4 for  $Re<sub>D</sub> = 3 \times 10^6$ .

Decreasing the Reynolds number results in higher gas temperatures since the transport of energy due to Joulean heating by the turbulent mechanism is reduced. Results are also presented in Fig. 3 for a magnetic damping function of unity in equation (8), that is the OHD form. The damping function for *Re,*   $= 3 \times 10^6$  and  $Ha = 200$  has a value of 0.96. The inclusion of the damping function produces higher gas temperatures since the turbulent viscosity is reduced resulting in less energy being transported to the wall. In comparison with temperatures evaluated for Hartmann MHD flow where laminar flow is assumed [5, 12, 131, the present temperature profiles are for significantly higher Hartmann numbers. Thus, by including turbulent transport in the energy equation, higher Hartmann numbers are possible with physically reasonable temperature profiles.

# *Radiating gas*

Representative temperature profiles for a radiatively participating gas are displayed in Fig. 4 for Reynolds and Hartmann numbers of  $3 \times 10^6$  and 200, respectively, and  $N = 0.001$ . Results are presented for channel optical thicknesses ranging from zero corresponding to a transparent gas to 50. Values for the other governing parameters are provided in the figure.



FIG. 5. Effect of conduction-radiation parameter.

As the opticai thickness increases from 0 to 10.0, significant decreases in gas temperatures are observed. For optical thicknesses less than about 0.1, the results are adequately described by those when the gas is assumed to be optically thin [16]. For optical thicknesses greater than 10.0, gas temperatures continue to decrease and become nearly insensitive to optical thickness particularly near the channel centerline. At very large optical thicknesses corresponding to an optically thick gas, the product of the divergence of the radiative heat flux and square of the channel optical thickness is independent of optical thickness [16]. Thus, the gas temperatures as observed by equation (3) become independent of the optical thickness. For the anticipated optical thickness range of 1 to 10, centerline gas temperatures vary from approximately 1.4 to 2.0 higher than the wall temperatures.

Effects of the conduction-radiation parameter on gas temperature profiles are illustrated in Fig. 5 where results are shown for values of the parameter equal to 0.0001, 0.001 and 0.01 with optical thicknesses of 1.0, 5.0 and 10.0. As the conduction-radiation parameter decreases, the radiative term in equation (3) becomes more significant with a resultant decrease in gas temperatures. The optically thick limit where gas temperatures are independent of optical thickness is attained at smaller values of the optical thickness for the Iower values of the conduction-radiation parameter.

The influence of the Reynolds and Hartmann numbers is demonstrated in Fig. 6 where gas temperature profiles are presented for  $N = 0.001$ ,  $\tau_0 = 5.0$ , and several values the Reynolds and Hartmann numbers. As expected, higher Hartmann numbers produce higher gas temperatures and higher Reynolds numbers yield lower profiles as a result of the increase turbulent transport. Gas temperatures are observed to be more sensitive to the higher values of the Reynolds number.

#### Nusselt numbers

In Table 1, results are tabulated for the bulk temperature, as well as convective, radiative, and total Nusselt numbers for various values of the governing parameters. The table is organized into four groups corresponding to temperature profiles presented in Figs. 3-6. The first group of results is for a transparent gas. The OHD flow produces a convective Nusselt number of 7310 for  $Re_b = 3 \times 10^6$ . Experimental measurements for OHD turbulent flow [26] yield a Stanton number correlation from which a Nusselt



FIG. 6. Effect of Reynolds and Hartmann numbers.

$ReD$ , 10 <sup>6</sup>	Ha	$\boldsymbol{N}$	$\tau_0$	$\theta_b$	$Nu_c$ , 10 <sup>4</sup>	$Nu_r$ , $10^4$	Nu, 10 <sup>4</sup>
3	$\bf{0}$	$\mathbf{0}$	$\bf{0}$	1.07	0.731	0.0	0.731
	150			4.17	0.139	0.0	0.139
$\begin{array}{c} 1 \\ 3 \\ 3 \end{array}$	150			2.30	0.395	0.0	0.395
	200			3.18	0.371	0.0	0.371
$\overline{\mathbf{3}}$	200	0.001	0.1	2.98	0.378	0.066	0.445
			0.2	2.70	0.391	0.183	0.574
			0.5	2.21	0.427	0.521	0.948
			1.0	1.87	0.475	0.978	1.45
			2.0	1.61	0.540	1.66	2.20
			5.0	1.41	0.626	2.86	3.49
			10.0	1.32	0.667	3.87	4.54
			20.0	1.27	0.680	4.76	5.44
			50.0	1.24	0.654	5.29	5.94
$\overline{\mathbf{3}}$	200	0.01	1.0	2.63	0.396	0.218	0.614
			5.0	2.09	0.434	0.652	1.09
			10.0	1.94	0.438	0.881	1.32
		0.0001	1.0	1.27	0.826	4.54	5.37
			5.0	1.08	1.98	17.4	19.4
			10.0	1.06	2.54	24.0	26.5
$\frac{1}{3}$	150	0.001	5.0	1.30	0.373	2.44	2.81
				1.27	0.732	2.40	3.13
10				1.22	2.13	2.36	4.49
$\mathbf{1}$	200			1.44	0.289	2.92	3.21
3				1.41	0.626	2.86	3.49
10				1.34	1.70	2.72	4.43

Table 1. Bulk temperatures and Nusselt numbers\*

\* Results are for  $K = 0.5$ ,  $Pr = 0.85$ ,  $Pr_t = 1.0$ ,  $Ec = 0.2$ .

number of 7100 is calculated for  $Re<sub>n</sub> = 3 \times 10^6$ . Thus, the results presented here for OHD appear reasonable. As the Hartmann number is increased, the convective Nusselt number decreases since the increase in temperature gradient at the wall is offset by a more rapid increase in bulk temperature. Results in the second group for a radiating gas illustrate that as the channel optical thickness increases, the convective Nusselt number increases and then decreases for optical thicknesses greater than about 20.0. The radiative Nusselt number continues to increase with optical thickness and shows a tendency of becoming independent of optical thickness which is characteristic of an optically thick gas. The radiative Nusselt number is the significant contributor to the total Nusselt number for optical thicknesses greater than 5.0. As the conduction—radiation parameter is increased, the Nusselt numbers presented in the third group exhibit large increases with optical thickness particularly for values of this parameter less than 0.001. Convective Nusselt numbers for values of the conduction- radiation parameter less than 0.01 arc nearly independent of the optical thickness. The Nusselt numbers for  $N = 0.001$  exhibit rather significant increases as the optical thickness is varied from 0.5 to 10.0 even though the corresponding gas temperature profiles in Fig. 5 and bulk temperatures in Table 1 change only slightly. The fourth group of results shows that as the Hartmann number is increased from 150 to 200, convective Nusselt numbers decrease but radiative Nusselt numbers increase, and total Nusselt numbers increase except for  $Re_b = 10^7$ where a small decrease is found. Increasing the Reynolds number from  $10^6$  to  $10^7$  produces significant increases in the convective Nusselt number, and small decreases in the radiative Nusselt number. These trends are consistent with the temperature levels and gradients in Fig. 6.

#### **CONCLUSIONS**

Analyses and results have been presented to examine the influence of gas radiation on temperature profiles for fully developed turbulent MHD flow. Representative models for turbulent gas velocity profiles as a function of the Reynolds and Hartmann numbers, turbulent viscosity with a magneticdamping function, current density, and radiative transfer for an absorbing and emitting gas with constant absorption coefficient were introduced. Analyses were also presented for gas bulk temperature as well as convective, radiative, and total Nusselt numbers. The results illustrated that gas radiation lowers significantly gas temperatures. Higher Hartmann numbers yield higher gas temperatures but higher Reynolds numbers produce lower temperatures as a result of the increased turbulent transport. The convective Nusselt number remains nearly constant as the optical thickness increased. At intermediate to large values of optical thickness, the radiative Nusselt number dominates the

convective Nusselt number and exhibits a strong dependency on optical thickness.

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### ECOULEMENT TURBULENT MHD D'UN GAZ RAYONNANT

Résumé—Les hautes températures et les produits gazeux de la combustion des hydrocarbures dans des générateurs MHD de grande taille nécessitent le développement de modèles précis pour la prédiction des profils de temperature des gaz et des flux thermiques parietaux. Aux temperatures elevees et pour les grandes tailles, le transfert thermique par rayonnement dans les gaz de combustion peut être le mécanisme principal de transfert d'energie. On presente des resultats sur les profils de temperature de gaz pour l'ecoulement MHD turbulent développé d'un gaz gris qui rayonne entre deux plans parallèles infinis avec un champ magnétique applique perpendiculairement aux plans. On port attention aux profils de temperature et au nombre de Nusselt pour différentes valeurs de nombres de Hartmann et de Reynolds, différentes épaisseurs optiques de gaz et pour le paramètre de couplage conduction-rayonnement.

## TURBULENTE MAGNETO-HYDRODYNAMISCHE STRÖMUNG EINES STRAHLENDEN GASES

Zusammenfassung-Die hohen Temperaturen und die gasförmigen Produkte bei der Verbrennung von kohlenwasserstoffhaltigen Brennstoffen in vorgeschlagenen großen magnetohydrodynamischen Generatoranlagen erfordern die Entwicklung genauer Modelle zur Berechnung der Temperaturprofile im Gas und der Warmestromdichten an der Wand. Bei den hohen Temperaturen und grogen Abmessungen wird der Strahlungswärmeaustausch in den Verbrennungsgasen ein bedeutender Wärmeübertragungsmechanismus sein. Es werden Ergebnisse fiir Temperaturprofile im Gas dargestellt fur den Fall der voll ausgebildeten turbulenten magnetohydrodynamischen Strömung eines grau strahlenden Gases zwischen zwei unendlichen parallelen Platten. Senkrecht zu den Platten wirkt das magnetische Feld. Besonderes Gewicht wird auf die Untersuchung von Temperaturprofilen und Nusselt-Zahlen fiir verschiedene Werte der Hartmann- und Reynolds-Zahl, der optischen Dicke des Gases sowie des Verhaltnisses zwischen Warmeleitung und Strahlung gelegt.

## ТУРБУЛЕНТНОЕ МАГНИТОГИДРОДИНАМИЧЕСКОЕ ТЕЧЕНИЕ ИЗЛУЧАЮЩЕГО  $\Gamma$ A3A

Аннотация - Высокие температуры и газообразные продукты сгорания углеводородных топлив в крупных магнитогидродинамических генераторах потребовали разработки точных моделей для расчета профилей температуры газа и тепловых нагрузок на стенке. При высоких температурах и больших размерах установок лучистый теплоперенос в газообразных продуктах сгорания может быть важным механизмом переноса энергии. Представлены результаты по определению профилей температуры газа при полностью развитом турбулентном магнитогидродинамическом течении излучающего серого газа между бесконечными параллельными пластинами, перпендикулярно к которым приложено магнитное поле. Особое внимание обращено на профили температуры, число Нуссельта при различных значениях чисел Гартмана и Рейнольдса, оптической **TOJII4HHbIra3aEinapaMeTpa,onHcbIBaIO~eroCOOTHOILIeHHe** *Mexny* **TeWIOupOBOaHOCTbK,k,R3JlySeHBeM.**